

## Supplementary Materials for

Analysis of the liquid column atomization by the annular dual-nozzle gas jet flow

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Movie S2. The atomization process of the falling water column depending on the momentum flux ratio of the liquid to the lower gas jet: (a, b)  $m_{L2} = 0.0056$ ; (c) 0.051; (d) 0.14. For all cases,  $We_{eff2} = 112$ ,  $Re_2 = 1.1 \times 10^5$ , and  $\theta_2 = 40^\circ$ .

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## Comparison of single-phase and two-phase gas velocities

Because of the presence of atomized droplets and liquid interface, it is expected that the gas velocity measurement will be affected by the shadows, strong light reflection from the interface and the relatively lower speed of droplet in the same image. Thus, it was not allowed to validate the gas velocity measurement at the regions where these influences are substantial. Thus, we evaluated the two-phase velocity fields outside of the liquid column (at  $x/D_2 > 0.2$ ) for  $m_{L2} = 0.009$ . [Figure S1](#) shows the profile of mean vertical velocity and r.m.s. horizontal velocity fluctuation along the streamwise location for  $m_{L2} = 0$  (without the liquid column) and 0.009 (with the liquid column). The operational conditions are  $m_{12} = 0.27$ ,  $\theta_2 = 20^\circ$ ,  $We_{eff1} = 13.6$ , and  $We_{eff2} = 52.0$ . It is clear that the mean and fluctuating gas velocities are not affected significantly by the presence of the liquid column. At the downstream locations at  $y/D_2 > 0.75$ , the liquid phase tends to decelerate the gas flow and increase the turbulence of the gas-phase flow slightly, but the primary atomization of the present configuration occurs at  $y/D_2 < 0.5$ , as explained. Based on this comparison, it is understood that the statistics of gas-phase outside of the liquid flow would not change significantly, despite the widespread of the spray along the horizontal direction.

## Estimation on the jet exit velocity

Due to the inherent limitation of PIV, the area near the jet exit could not be measured with PIV (section 2.2). However, the jet exit velocity is necessary to calculate  $Re$ ,  $We$  and momentum ratio to define the characteristics of the jet. Here, the exit velocity of the jet is estimated using the particle tracking velocimetry and the conservation of momentum at the jet exit. From the [equation \(4.2\)](#) and using the control volume  $C_2$  in [figure 14](#), the following relationship can be deduced when  $u_2 = 0$ :

$$\widehat{v}_{4G}^2 A_4 = u_1^2 A_1. \quad (S1)$$

From this equation, the exit velocity of upper nozzle ( $u_1$ ) can be theoretically calculated as:

$$u_1 = \left( \widehat{v}_{4G}^2 A_4 / A_1 \right)^{1/2} = \left( \int_{S_3} v_{3i} (v_{3i} \cdot \bar{n}) dA \cdot A_4 / A_1 \right)^{1/2}. \quad (S2)$$

Likewise, the exit speed of the upper nozzle ( $u_2$ ) can be calculated, substituting  $u_1 = 0$ :

$$u_2 = \left( \widehat{v_{4G}}^2 A_4/A_1 \right)^{1/2} = \left( \int_{S_3} v_{3i} (v_{3i} \cdot \bar{n}) dA \cdot A_4/A_1 \right)^{1/2}. \quad (S3)$$

Note that the exit velocity  $u_1$  and  $u_2$  are the function of vertical location ( $y$ ) of  $S_3$ . The experiment data are applied to the equations (S2) and (S3). And the result for the upper and the lower jet is shown in figures S2(a) and (b), respectively. For all velocities, it becomes constant after  $y/D = 0.6$ , and gradually decreases after that. Here, we infer the jet exit velocity by averaging the values for  $y/D = 0.6-1.0$ . On the other hand, the value obtained by using particle tracking velocimetry at the jet exit is displayed on the y-axis in open symbols. In order to predict the jet outlet velocity under various pressure conditions, the jet outlet velocities versus applied pressure are plotted (figure S3). Here, it was experimentally found that the jet outlet velocity is proportional to the third power of the applied pressure. Through this, the following was obtained.

$$u_{1,2} = c \cdot p_{1,2}^{1/3}. \quad (S4)$$

The proportional constant ( $c$ ) for the upper nozzle and the lower nozzle could be 240 and 145, respectively. Through this equation, the jet exit velocity under various experimental conditions was inferred and the main dimensionless number ( $Re$ ,  $We$ , momentum flux ratio) was calculated.

### PDF of the atomized droplet size

Figure S6 shows representative PDF distribution of the droplet size. To obtain statistically valid data for droplet diameter, the total number of droplets should be sufficient. As seen in the figure, the distribution of droplet size becomes converged after  $N > 500$ , while the number of detected droplets is larger than 10,000. Thus, we thought that the PDF is converged and the averaged droplet size is appropriate for further analysis. In addition, the PDF distribution follows the gamma distribution due to the size polydispersity of liquid breakup as discussed by Villermaux *et al.* (2004) and Bremond *et al.* (2007).

### Convergence of the measured droplet size

First, we estimated the characteristic time scale for the liquid breakup using the Rayleigh-Taylor instability, which governs the disintegration process in this study. From the equation

(4.10) in the main manuscript, the maximum growth rate ( $\omega_{i,max}$ ) determines the breakup time ( $T_b = 2\pi/\omega_{i,max}$ ) and is calculated as below (Marmottant & Villermaux 2004).

$$\omega_{i,max} = \left(\frac{2}{3\sqrt{3}}\right)^{1/2} \left(\frac{\rho_l a^3}{\sigma}\right)^{1/4}.$$

Here,  $\rho_l$ ,  $a$ , and  $\sigma$  is the liquid density, acceleration of interface, and the coefficient of surface tension, respectively. Using the interfacial acceleration estimated in equation (4.11), the growth rate is obtained as 15.2 - 1809.3 m/s<sup>2</sup>, corresponding to  $\omega_{i,max}$  of 52.3 - 1881.7 1/s. Thus, the breakup time-scale is estimated as  $T_b = 0.0033 - 0.12$  sec. For the atomized droplet size measurements, we obtained 300 images at the speed of 200 Hz, which means the total measurement time ( $T_m$ ) of 1.5 sec. In figure S7, we have plotted the variation of droplet size ( $d_{32}$ ) with  $T_m/T_b$ , for selected cases. The cases were chosen for the higher possibility of not achieving the convergence. As shown, the droplet size seems to be converged statistically for the tested cases.

### Flapping frequency of the liquid column

The successive images of high-resolution (500 × 1000 pixels) were used to measure the dominant frequencies using the fast Fourier transform. The capturing rate ranges as 1,710-10,300Hz, indicating that the maximum measurable frequency is 805-5,150Hz by the Nyquist sampling theorem. Knowing that the reported frequencies of the liquid column flapping in previous studies (Marmottant & Villermaux 2004), this measurement range is sufficient for extracting the most dominant frequency representing the present unsteadiness. To obtain the dominant frequencies, the images are binarized with the specific threshold so that the edge of the liquid phase is clearly distinguished from the background. At a given vertical location, the horizontal center of the liquid area is collected (the location is drawn in figure S8(a) as a dashed line) for all times, and the fast Fourier transform is applied to find the magnitude of each frequency at every vertical location (figure S8b).

Figure S9(a) shows the variation of the dominant frequency ( $f_d$ ), depending on the lower nozzle angle ( $\theta_2$ ), for the operation conditions of  $m_{L2} = 0.027 - 0.051$ ,  $We_{eff2} = 28 - 468$ , and  $m_{12} = 0 - 0.27$ . The measured frequency is in the same range of previous studies (Marmottant & Villermaux 2004; Tian *et al.* 2014). For the present data, it is observed that the most dominant

frequency decreases with increasing of  $\theta_2$ . The low frequency ( $< 50$  Hz) at  $\theta_2 > 40^\circ$  is attributed to the wake-like flow structure made by the lower jet with the steep angle while the fast oscillatory behavior for  $\theta_2 = 20^\circ$  is attributed to the resemblance to the coaxial flow. On the other hand, it is observed that  $m_{12}$  and  $f_d$  are weakly correlated, which is possibly because the upper jet becomes strong enough to alter the global flow from the wake-like to the coaxial-like flows.

To examine the relation between the flow regime and the frequency, the dominant frequency is plotted at each flow regime in [figure S9\(b\)](#). It is clear that the flow regime having the back-flow structure is characterized by the lower frequency ( $< 50$ Hz) since the wake-like behavior of gas-phase slows down the liquid oscillation. In contrast, if the upper jet overcomes the back flow and the flow regime becomes the bulk or droplet atomization regime, the flow structure has a high frequency ( $\sim 100$ Hz) due to the coaxial-like behavior of the gas phase. The trend of frequency also provides quantitative evidence for flow regime classification between the back-flow and the non-backflow regimes.

### **Effect of liquid interface on the gas velocity: control volume analysis**

To estimate the liquid-induced deviation in the mean gas velocity for all locations, we need the spatial distribution of velocity, size, and volume fraction of dispersed phase, with varying conditions, which is unfortunately not allowed in the present setup. However, our major interest is focused on the primary atomization site near the nozzle (usually  $y/D_2 > -0.5$ ), and we tried to estimate the two-phase flow near the upper-nozzle section,  $-0.5 < y/D_2 < 0.5$ , where the upper jet and the liquid column interact. In [figure S10](#), we have illustrated the control volume ( $C_3$ ) to analyze this. Here, the upper gas jet is generally dragged by the lower-speed liquid flow and the gas velocity at the interface is the minimum. The interfacial velocity is modeled as  $u_i = (\sqrt{\rho_L}u_L + \sqrt{\rho_G}u_1)/(\sqrt{\rho_L} + \sqrt{\rho_G})$  assuming that the dynamic pressure balances through the transverse direction ([Dimotakis 1986](#)), and is calculated as  $u_i/u_1 = 0.045 - 0.055$ . The thickness of this low-speed region (relevant to the shear-layer thickness) is obtained as the vorticity length,  $\delta_\omega = 0.17(1 - r)/(1 + r) \cdot l_m$ , where  $r = u_1/u_L$  and  $l_m$  denotes the streamwise length of the mixing layer.  $\delta_\omega/D_2$  is calculated as  $0.081 - 0.083$  at the lower nozzle exit ( $y = 0$ ). We assumed that the shear layer evolves linearly from the liquid surface expanding horizontally about half the thickness of the vorticity layer ( $\delta_\omega/2$ ). Thus, the net gas velocity ( $\hat{v}'_{3G}$ )

decelerated by the interaction with the liquid column can be calculated by modifying the equation (4.3) in the main text, as follows.

$$\rho_G \hat{v}'_{3G} | \hat{v}'_{3G} | A_{3G} = \int_{S_3} p_{3G} dA - \rho_G v_1^2 A_1 - \int_{S_4} \tau_s dA. \quad (S5)$$

Here,  $S_4$  denotes the interface between gas and liquid with the velocity of  $u_i$ , and the frictional stress ( $\tau_s$ ) there can be expressed as  $\tau_s \approx \mu(u_1 - u_i)/\delta_\omega$ , where  $\mu$  is the kinematic viscosity of the gas. Using the representative value of  $u_i/u_1 = 0.05$  and  $0.5\delta_\omega/D_2 = 0.04$ , the contribution by the viscous stress is calculated as  $\int_{S_4} \tau_s dA \approx \mu(u_1 - u_i)/\delta_\omega \cdot \pi D_L h_2$ . Compared to the single-phase gas flow contributed by the momentum of the upper jet, it is estimated that the deceleration of  $\hat{v}'_{3G}$  is approximately 1.68% (below 2%), supporting our original assumption, i.e., the analysis with the single-phase (gas velocity) only is reasonable.

### Examination on the transition of breakup mechanism

We examined whether the transition from the Rayleigh-Taylor (RT) to the shear-induced Kelvin-Helmholtz (KH) instability occurs in the present cases. According to [Theofanous \(2011\)](#), this transition starts at  $We_d \sim 10^2$  ( $d$ , drop diameter) and completes at  $We_d \gtrsim 10^3$  (at low  $Oh < 0.1$ ).  $Oh$  is the Ohnesorge number. At this transition regime, the ratio between the droplet size and the instability length scale, i.e.,  $d/\lambda_{KH}$ , determines the RT-to-KH transition, and its critical value is  $\sim 10$  ([Sharma et al. 2021](#)). The regime map for RT and KH instability is shown in [figure S11](#), reproduced from [Sharma et al. \(2021\)](#). For the droplet atomization regime, the droplet pre-atomized from the upper jet experiences another breakup with the lower nozzle. Looking closely, the droplets are affected by the high-speed freestream and the shear layer with the diffusion length occurs on its surface at the windward side. Using the dispersion relation ([Marmottant & Villermaux 2004; Kim et al. 2006; Sharma et al 2021](#)), the length scale of shear-induced instability ( $\lambda_{KH}$ ) is obtained as  $\lambda_{KH} = 2\pi/1.5 \cdot \delta_a \sqrt{\rho_l/\rho_g}$  where  $\rho_{l,g}$  denote the density of liquid and gas, respectively. Here,  $\delta_a$  is the boundary-layer thickness of the air phase and can be determined using boundary-layer analysis ([Kim et al. 2006; Jalaal & Mahravaran 2014](#)) as  $\delta_a = K \cdot l_c / Re_{l_c}$  where  $l_c$  corresponds to the characteristic length and is set to  $d/2$  ([Sharma et al. 2021](#)). The Reynolds number ( $Re_{l_c}$ ) is defined as  $\rho_g u_i l_c / \mu_g$ . The scaling constant ( $K$ ) is set to 0.12 given from the experiment and simulation of droplet differentiation

(Kim *et al.* 2006; Jalaal & Mahravarán 2014; Sharma *et al.* 2021), and the effective gas velocity ( $u_{eff}$ ) is used as a characteristic velocity ( $u_i$ ) like the estimation of  $We_{eff}$  (see §2.4). The Weber number ( $We_d$ ) can be calculated as  $We_d = \rho_g u_{eff}^2 d / \sigma$  ( $\sigma$ : the surface tension). Here, we used the droplet size ( $d$ ) when the lower nozzle is absent (figure 20) since the fragmentation process of the liquid column at the upper gas jet ( $y > 0$ ) is less perturbed by the lower gas jet. Among the cases in the regime map (figure 13), three conditions ( $m_{12} = 0.27$  and  $\sin\theta_2 = 0.34 - 0.64$ ) belong to the droplet atomization regime, and their length ratios and Weber numbers were plotted onto the figure S11. The figure shows that our cases mainly lie on the RT regime, while the case of the highest  $We$  ( $= 115$ ) is on the border of the transition regime. The Low  $We$  ( $\lesssim 100$ ) indicates the low dynamics pressure which inhibits the KH instability to grow and gives sufficient time for slowly-growing RT instability (Theofanous 2011; Theofanous *et al.* 2012). Additionally, the length ratios are below the threshold ( $d/\lambda_{KH} = 10$ ). Since the shear-induced instability only develops at the sufficiently high wavenumber (i.e., high  $d/\lambda_{KH}$ ), the  $\lambda_{KH}$  comparable to the droplet diameter ( $\sim 1.23d$ ) in the present study also obstructs the growth of the KH instability. It should be noted that, if the  $We_d$  increases further than our range (e.g., by increasing applied pressure in Table 1), the KH instability should be eventually triggered, as expected from figure S11.

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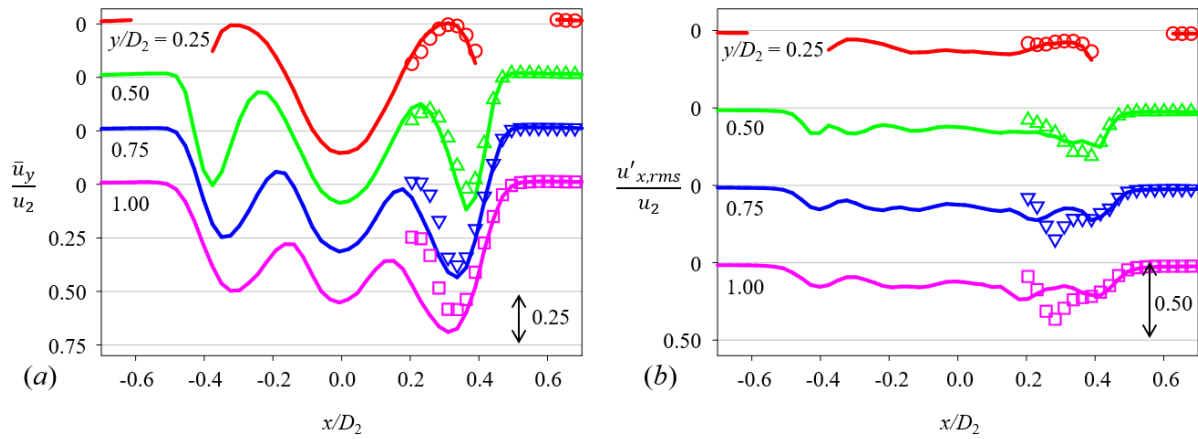


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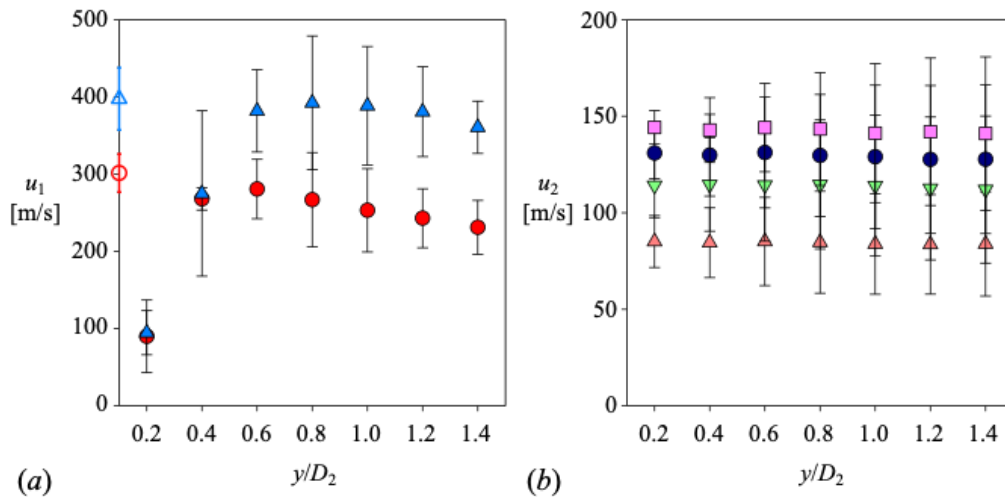


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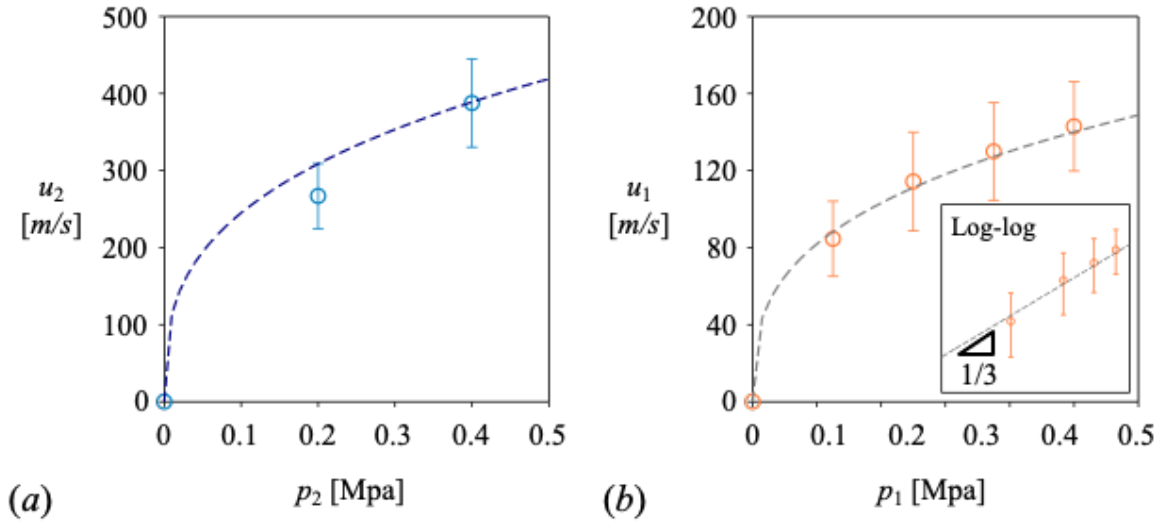


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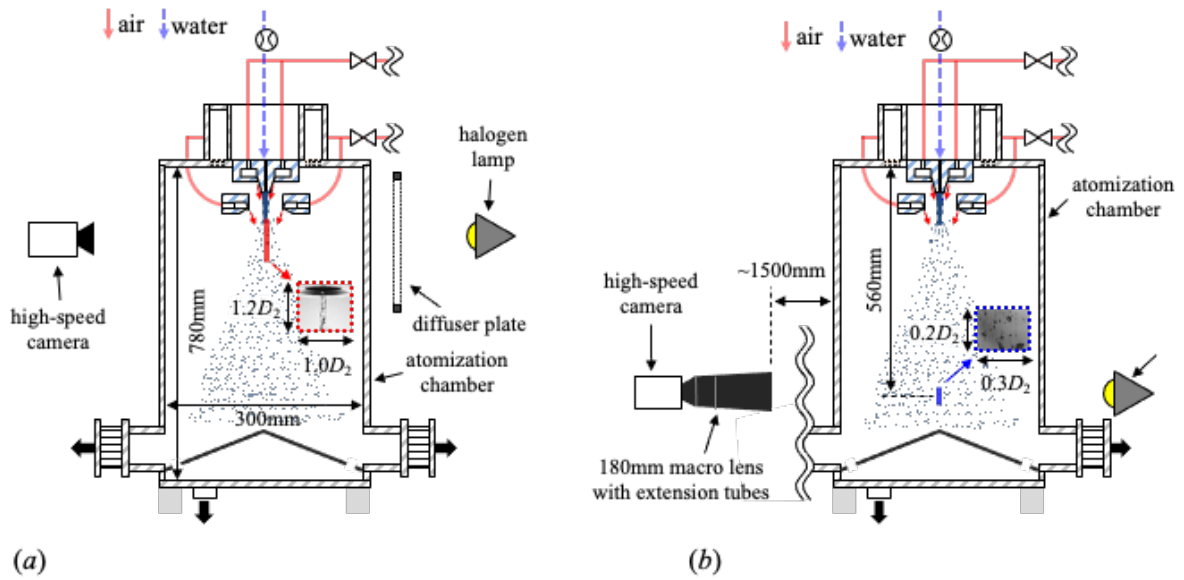


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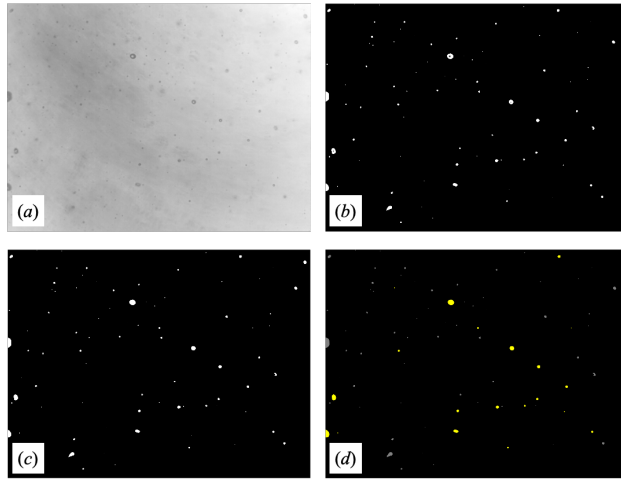


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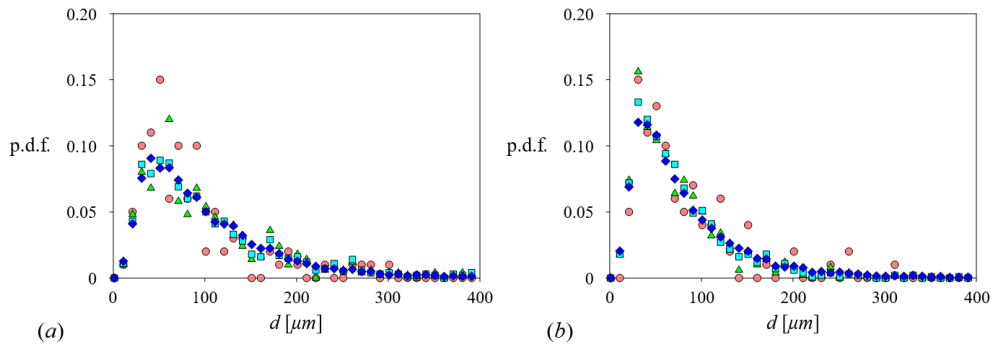


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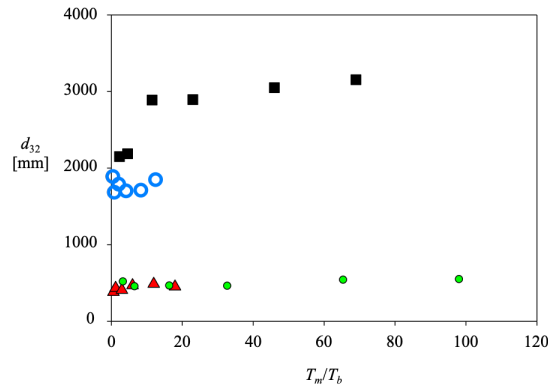


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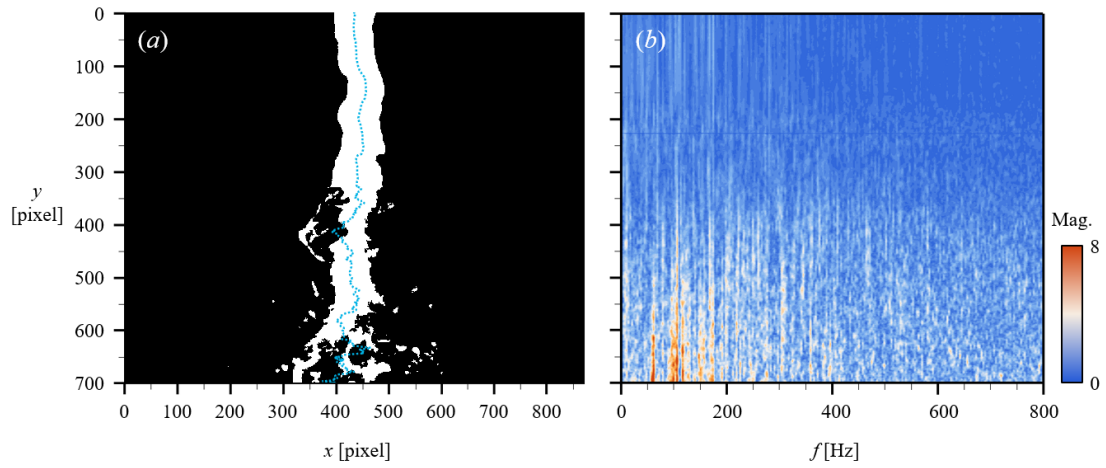


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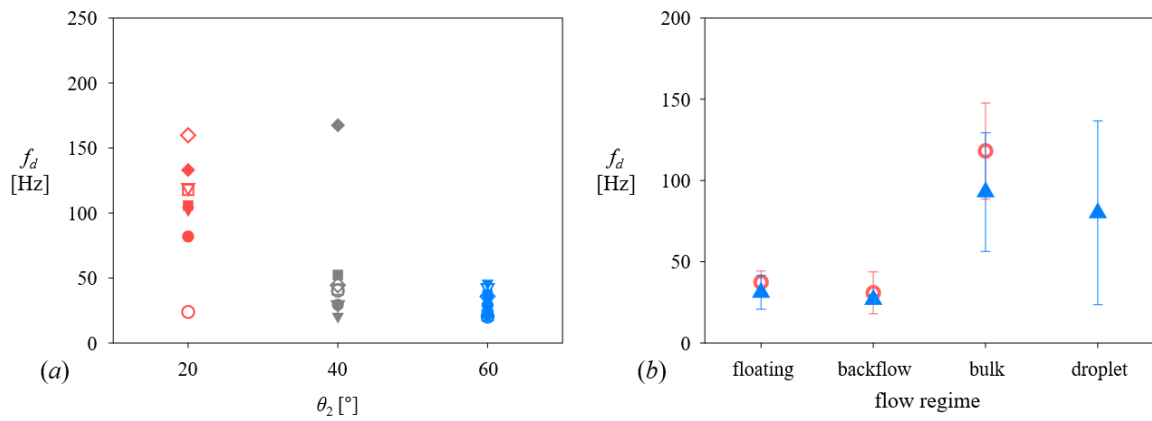


Figure S9. The dominant frequency depending on (a) the nozzle angle ( $\theta_2$ ) and (b) the flow regime: open symbol,  $m_{L2} = 0.051$ ,  $We_{eff2} = 28 - 252$ ; close symbol,  $m_{L2} = 0.027$ ,  $We_{eff2} = 52 - 468$ . In (a),  $\circ$ ,  $m_{12} = 0$ ;  $\nabla$ , 0.11;  $\square$ , 0.17;  $\diamond$ , 0.27.

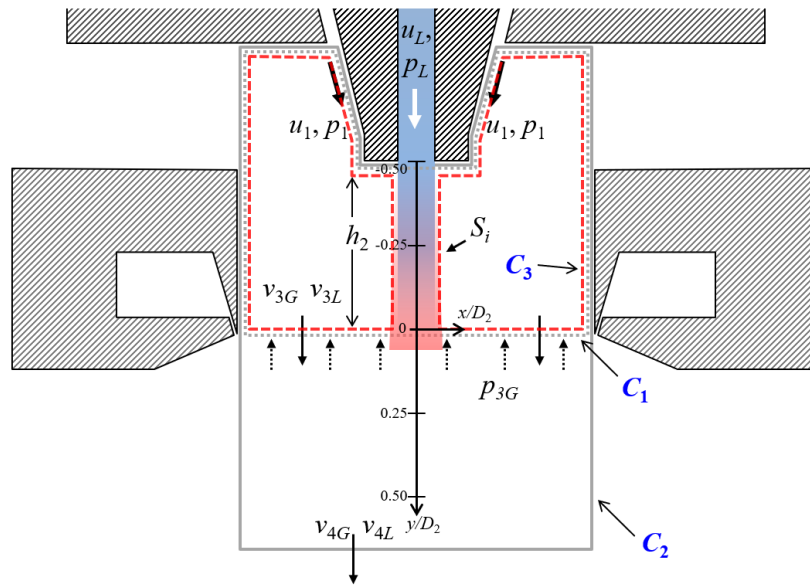


Figure S10. Control volume analysis to consider the interfacial friction owing to the presence of liquid lump. Here,  $S_i$  denotes the interfacial surface between the liquid lump and gas flow.

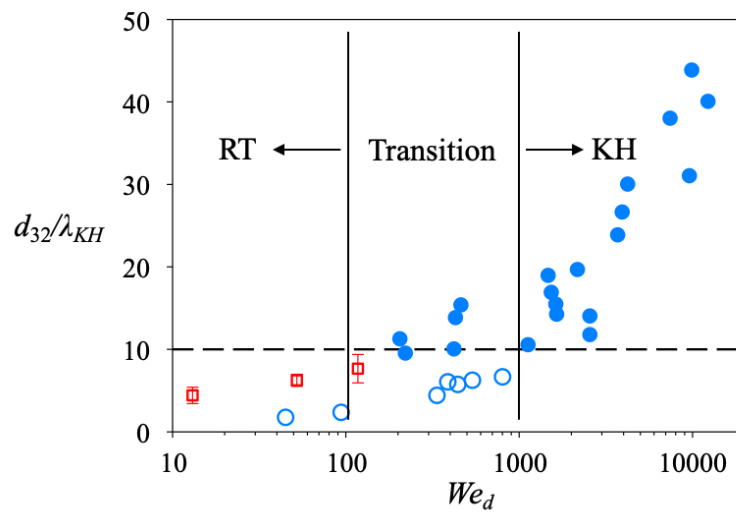


Figure S11. The length ratio ( $d_{32}/\lambda_{KH}$ ) depending on the Weber number for the droplet atomization regime ( $\square$ ). The circle symbols ( $\circ$ : Rayleigh-Taylor (RT) regime and  $\bullet$ : Kelvin-Helmholtz (KH) regime) are from [Sharma \*et al.\* \(2021\)](#).